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SID 62-219

BOILER PLATE NO. 15, SPACECRAFT SATURN RESEARCH AND DEVELOPMENT TEST SA-6 SPECIFICATION (Unclassified) NAS 9-150

28 February 1962

4.5.2.3



Approved by

Vice President and Apollo Program Manager

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NORTH AMERICAN AVIATION, INC. SPACE and INFORMATION SYSTEMS DIVISION







CONTENTS

Paragraph	Title						Page
1.	SCOPE	•	•	•	•		1
1.1	Scope	•	•	•	•	•	1
2.	APPLICABLE DOCUMENTS .						1
2.1	General	•	•	•	•	•	1
		•	•	•	•	•	-
3.	REQUIREMENTS		•		•	•	1
3.1	General		•		•	•	1
3.2	Components	•		•	•	•	1
3.2.1	Arrangement			•	•		1
3.2.1.1	Command Module			•	•	•	3
3.2.1.2	Service Module .	•	•		•		3
3.2.1.3	Adapter Prototype	•		•	•		3
3.2.1.4	Launch Escape Tow	er P	rotot	ype	•		3
3.2.2	Equipment		•	•		•	3
3.3	Performance	•	•	•	•		3
3.3.1	General	•	•	•	•		3
3.4	Design and Construction .				•		4
3.4.1	General	•	•		•		4
3.4.2	Weight					•	4
3.4.3	Heat Shielding Simulation	n .		•		•	4
3.5	Ground Support Equipment.	•	•	•	•	•	4
3.5.1	Ground Handling Equipm	ent	•	•	•	•	4
4.	QUALITY ASSURANCE PROVISION	SNC	•	•	•		4
4.1	General			•		•	4
4.2	Inspections and Tests	•	•	•	•	•	4
5.	PREPARATION FOR DELIVERY		•	•	•	•	4
5.1	Airborne Equipment		•				4
5.2	Transportation		•			•	4
6.	NOTES	•		•		•	5
4 1	Definitions						5



APPENDICES

Appendix			Page
1-a	Government-Furnished Property, Contractor-		
	Installed, Communication and Instrumentation .	•	6
1-b	Contractor-Furnished Equipment, Contractor-		
	Installed, Communications and Instrumentation.	•	7
1-b	Contractor-Furnished Equipment, Contractor-		
	Installed, Power System.	_	8

ILLUSTRATIONS

Figure	Page
1	2





1. SCOPE

1.1 Scope. - This specification covers the requirements for the Spacecraft Saturn Research and Development Test SA-6 Boilerplate. The boilerplate shall provide demonstrations and obtain information relative to the flight characteristics of the Apollo Spacecraft and Saturn booster.

2. APPLICABLE DOCUMENTS

2.1 General. - The following documents shall form a part of this specification:

Government Documents

Air Force

ARDCM-80-1, Volume 1

Handbook of Instructions for

Aircraft Designers

National Aeronautics and Space Administration

NCP200-2

Quality Assurance Provisions

for Space Contractors, dated

15 December 1961

Space and Information Systems Division, North American Aviation, Inc.

SID 62-240

Preparation for Delivery of Airborne Equipment, General Requirements For

3. REQUIREMENTS

3.1 General. - The external configuration of the boilerplate shall be similar to the configuration of the prototype Apollo Spacecraft. The configuration of the boilerplate is shown in figure 1.

3.2 Components. -

3.2.1 Arrangement. - Equipment and ballast will be arranged in the boilerplate to simulate the center of gravity of the prototype Spacecraft. The boilerplate shall include the following:





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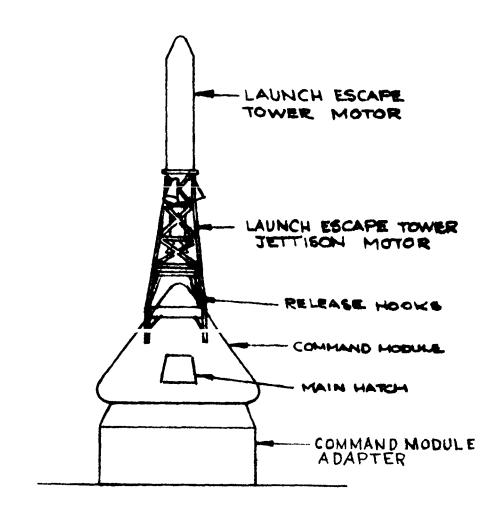


Figure 1. Boilerplate Number 15



COMPRESENTING

- (a) Command Module
- (b) Service Module
- (c) Adapter
- (d) Escape Tower
- 3.2.1.1 Command Module. The command module of the boilerplate shall include provisions for mating to the service module, and attachment of the launch escape tower. Provisions for separation of the spacecraft from the booster shall not be required. The command module shall include:
 - (a) Communications and Instrumentation System
 - (b) Environmental Control System (as required for system cooling)
- 3.2.1.2 Service Module. The service module shall include provisions for mating to the adapter, and the command module. The service module shall include:
 - (a) Communications and Instrumentation System
 - (b) Environmental Control System (ground cooling only)
 - (c) Battery Power Supply System
- 3.2.1.3 Adapter Prototype. The adapter prototype shall include provisions for mating to the service module boilerplate, and the saturn booster. The adapter prototype shall contain instrumentation.
- 3.2.1.4 <u>Launch Escape Tower Prototype</u>. The launch escape tower prototype shall have provisions for attachment to the command module of the boilerplate.
- 3.2.2 Equipment. The equipment for the boilerplate shall include the equipment listed in appendices I-A and I-B.
 - 3.3 Performance. -
- 3.3.1 General. The boilerplate shall simulate the dynamic characteristics of the final Appollo Spacecraft. The boilerplate shall be a research and development test vehicle and shall be tested on the Saturn SA-6 launch vehicle. The boilerplate shall:
 - (a) Demonstrate dynamic compatibility of spacecraft and the C-l boost system.







- (b) Provide structural demonstration for evaluation of flight loads, vibration, accelerations and temperature.
- (c) Evaluate the capability of obtaining orbital altitude range: 90 to 400 nm.

3.4 Design and Construction. -

- 3.4.1 General. ARDCM-80-1, Volume 1, shall be used for guidance and reference material in the design and construction of the boilerplate
- 3.4.2 Weight. The boilerplate shall have a mass and center of gravity similar to the mass and center of gravity of the prototype Apollo Spacecraft.
- 3.4.3 Heat Shielding Simulation. The boilerplate shall not contain heat shielding. The stiffness of the heat shielding of the spacecraft shall be simulated in the boilerplate.

3.5 Ground Support Equipment. -

3.5.1 General. - Ground Support equipment shall be required to transport and test the boilerplate. The requirement for ground support equipment are not part of this specification.

4. QUALITY ASSURANCE PROVISIONS

- 4.1 General. Quality Assurance Provisions for the boilerplate shall be in accordance with the applicable portions of NASA Bulletin NCP200-2.
- 4.2 Inspections and Tests. Inspections and tests to determine conformance of the boilerplate to contract and specification requirements shall be conducted prior to submission of the boilerplate to NASA or in the presence of an NASA representative. Results of inspection tests on major components shall be submitted to NASA for review. Other acceptance test data relative to this specification shall be maintained and made available for review to NASA upon request.

5. PREPARATION FOR DELIVERY

- 5.1 Airborne Equipment. Airborne Equipment shall be prepared for delivery in accordance with Specification SID 62-240.
- 5.2 <u>Transportation</u>. The boilerplate shall be delivered to the local area testing site.





6. NOTES

6.1 <u>Definitions</u>. - A boilerplate is a simulated spacecraft module for pre-developmental and/or developmental tests leading to the design of a prototype module.



Appendix I-a

Government-Furnished Property, Contractor-Installed

	Part No.														
Communication and Instrumentation	Description	Sensors	Recorders	Telemetry Systems	VHF Transmitters	Command Receiver/Lecoder	C-Band Beacons	Minitrack Beacon	S-Band Beacon	RF Multiplexer	UHF Z-Match	C-Band Antenna Elernents	Coaxial Switch	Power Dividers	Power Source
	Quantity	l Set	7	4	4	J	73	1	1	1	1	4	1	73	-
	Item No.		7	8	4	ß	9	7	œ	6	10	11	12	13	14





Appendix I-b

Contractor-Furnished Equipment, Contractor-Installed

e	Part No.			
Communications and Instrumentation	Description	RF Transmission Line	Slot Antennas	S-Band Antenna Elemerts
	Quantity	1	∞	4
	Item No.	-	2	3





Appendix I-b

Contractor-Furnished Equipment, Contractor-Installed

No.	
H	ļ
Pa	l

Description

Power Systern

Quantity

Item No.